**The report and documentation of ‘For Rocket’**

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This documentation is written to explain and guide for utilize of ‘For Rocket’ program. This program is basically designed and made to calculate the launch of rockets and simulate the launch itself including plenty of situations, information, and knowledge such as rocket mechanics, mathematics, physics, and astrophysics with ‘Python’ and ‘Tkinter’ as main and central tools. I totally want that this program can provide and allow tons of students, undergraduates, graduates, and other general people as well as scientists and engineers the invaluable and priceless opportunities to enjoy more over the space science and to open their myriad and infinite possibilities. I will account for a number of equations, terms, and symbols with the ones which are employed in mathematics, physics, and programing in common.

1. The basic rocket equation

First of all, this chapter requires readers to understand some knowledge with regarding to the rocket mechanics from basic things to profound things as an essential requisite.

1. Projectile motion

The projectile motion chapter is that examining and researching the object’s motions which are launched with some degree. Furthermore, we can make plots and tables(charts) with ‘matplotlib’ and ‘pandas’.

* Arrangement of terms in ‘Projectile motion equation’

>v0 : initial velocity. [m/s]

>deg : degree(has radian value) -> I will utilize this constant as ’θ’

>g : gravity of Earth. The value is 9.81 m/s^2

>v\_x = v0 \* cos(θ) [m/s] : the velocity of x-axis at initial time, which is comparable with the velocity of x-axis over time when other obstacles or resistances are not existed. This is because the velocity of x-axis is not affected by other accelerations, which means that the velocity of x-axis is not changed whenever the time flows.

>t\_h = v0 \* sin(θ) / g [s] : the time when the object reaches at the maximum height. This equation is made the cross with initial velocity(v0) and sin(θ), and this value is divided with gravity of Earth(g).

>h = (v0 \* sin(θ) \* t\_h) – (0.5 \* g \*(t\_h)\*\*2) : maximum height which the object can reach. This equation is composed with **1)** the multiplication of the v0(initial velocity), sin(θ), and t\_h(the time when the object reaches at the maximum height) and **2)** 0.5, g(gravity of Earth), and the square of t\_h. Then subtracting the equation **2)** from **1).**

1. Geostationary orbit
2. Escape speed(velocity)